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iru#  
Olkwz hlkw#Vsdhf0Edvng#R swlfv#  
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F ruqhuwrqh#Jhvdufk#J urxs/#qf1  
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D u#I ru fh#Jhvdufk#0deruwrul 4;#D xjxw#5337

*Applying Tomorrow's Materials Today*



## **ACKNOWLEDGMENT**

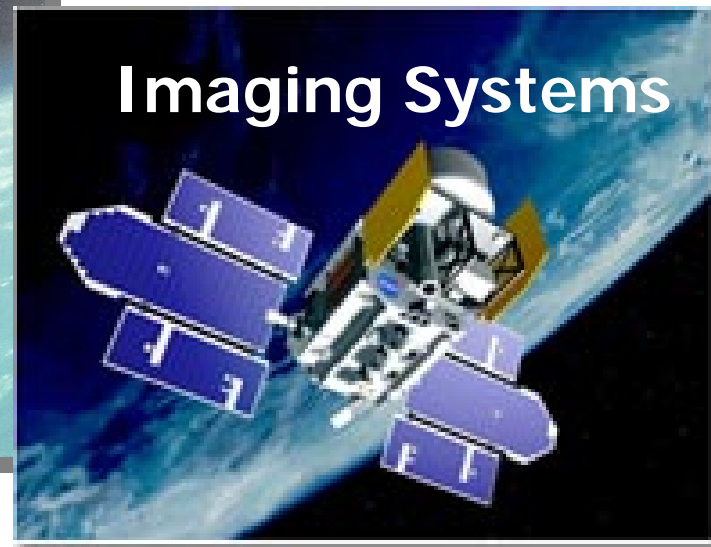
This presentation summarizes results of Small Business Innovation Research (SBIR) Phase I contract F33615-02-M-5027 (4/9/02 - 11/7/02) and effort to date for Phase II contract F33615-03-C-5013 funded by the Air Force Research Laboratory (AFRL) and managed by Dr. David Mollenhauer (AFRL/MLBC).

- Program Introduction
- Phase I Objectives
- Phase I Results
- Phase II Plan
- Phase II Early Results
- Phase II Enhancement
- Summary

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# PROGRAM INTRODUCTION

- Applications: Space-Based Optics



- Operational Need:  
Improve on glass & metal mirrors
  - Lighter
  - Tougher
  - Cheaper

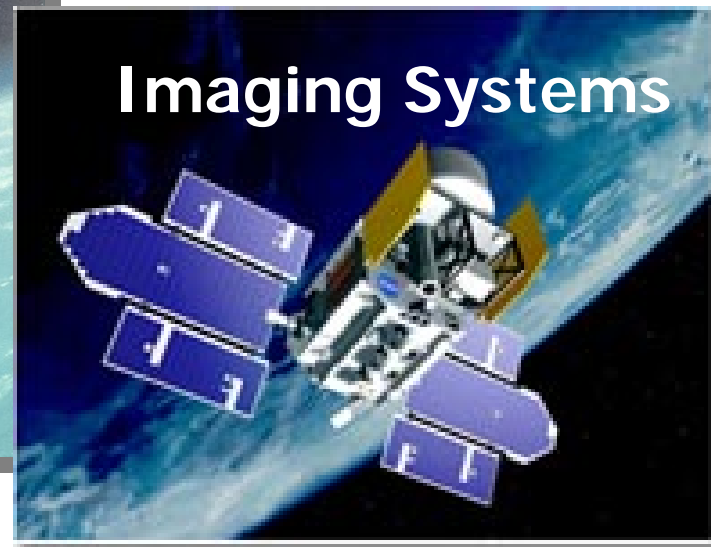
Images

L: [www.fas.org/spp/starwars/program/sbl.htm](http://www.fas.org/spp/starwars/program/sbl.htm)

R: [www.ball.com/aerospace/products/bus.html](http://www.ball.com/aerospace/products/bus.html)

## PROGRAM INTRODUCTION

- Applications: Space-Based Optics



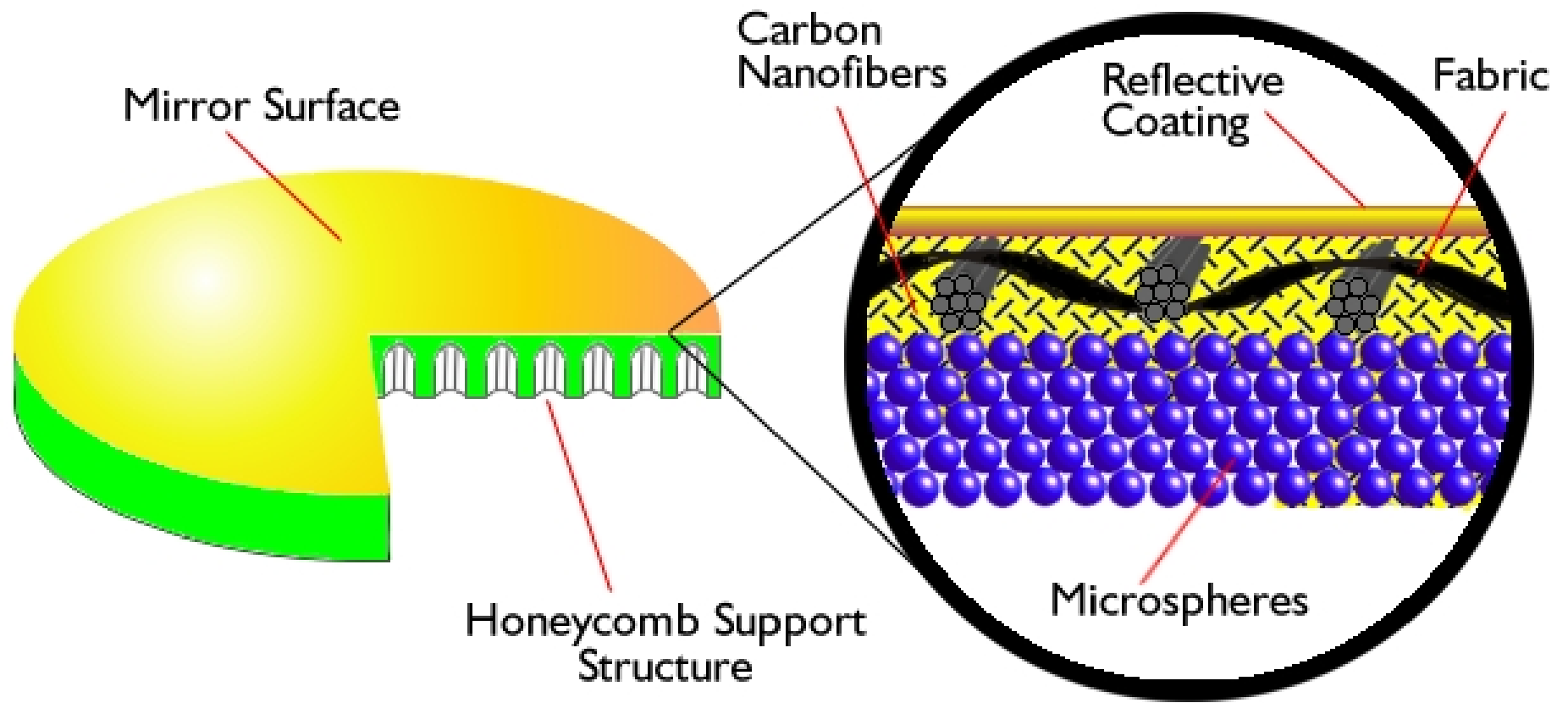
- Operational Need:  
Improve on glass & metal mirrors
    - Lighter
    - Tougher
    - Cheaper
- new materials
- new processes

## **PROGRAM INTRODUCTION: Material Design Elements**

- **Space compatible:**
  - Radiation hard (to space ambient)
  - AO resistant  
(inherent or through practical coating)
  - Resistant to out-gassing in vacuum
- **Improvement over glass or metal mirrors:**
  - Lower areal density
  - Higher tolerance to thermal excursion (low CTE)
  - Improved strength (toughness & stiffness)
- **Compatible with obtaining optical surface**

# PROGRAM INTRODUCTION: Material Concept

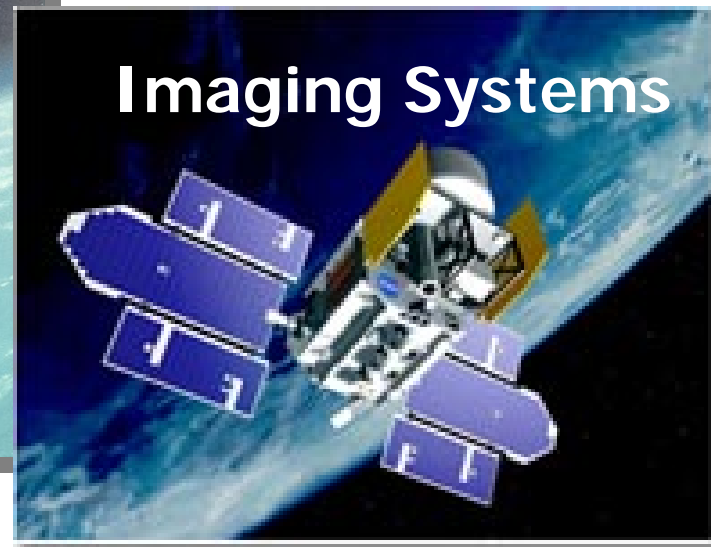
## Multi-Component Composites





## PROGRAM INTRODUCTION

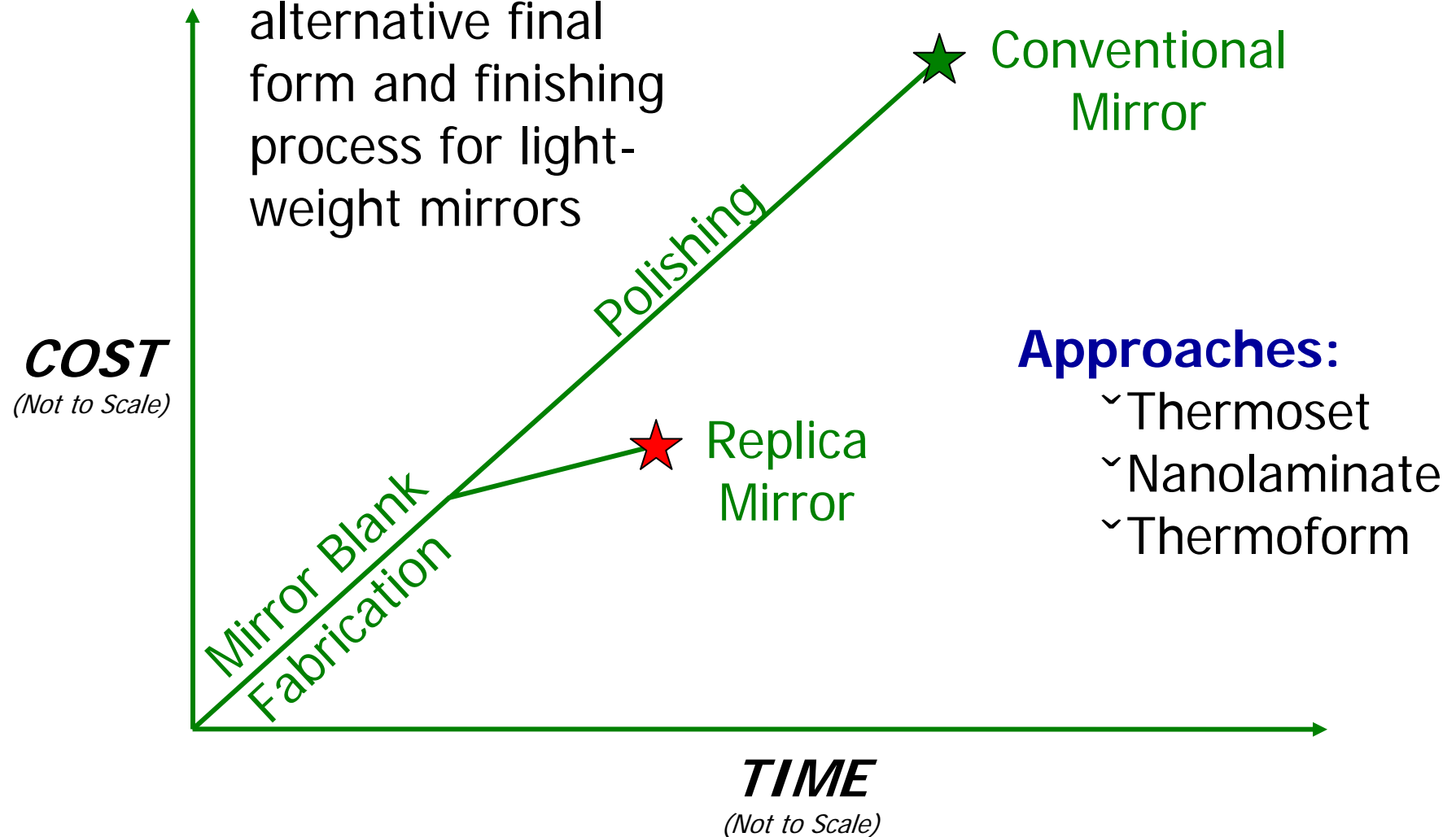
- Applications: Space-Based Optics



- Operational Need:  
Improve on glass & metal mirrors
    - Lighter
    - Tougher
    - Cheaper
- new materials
- new processes

# PROGRAM INTRODUCTION: Replication Technology

**Goal:** Develop alternative final form and finishing process for light-weight mirrors

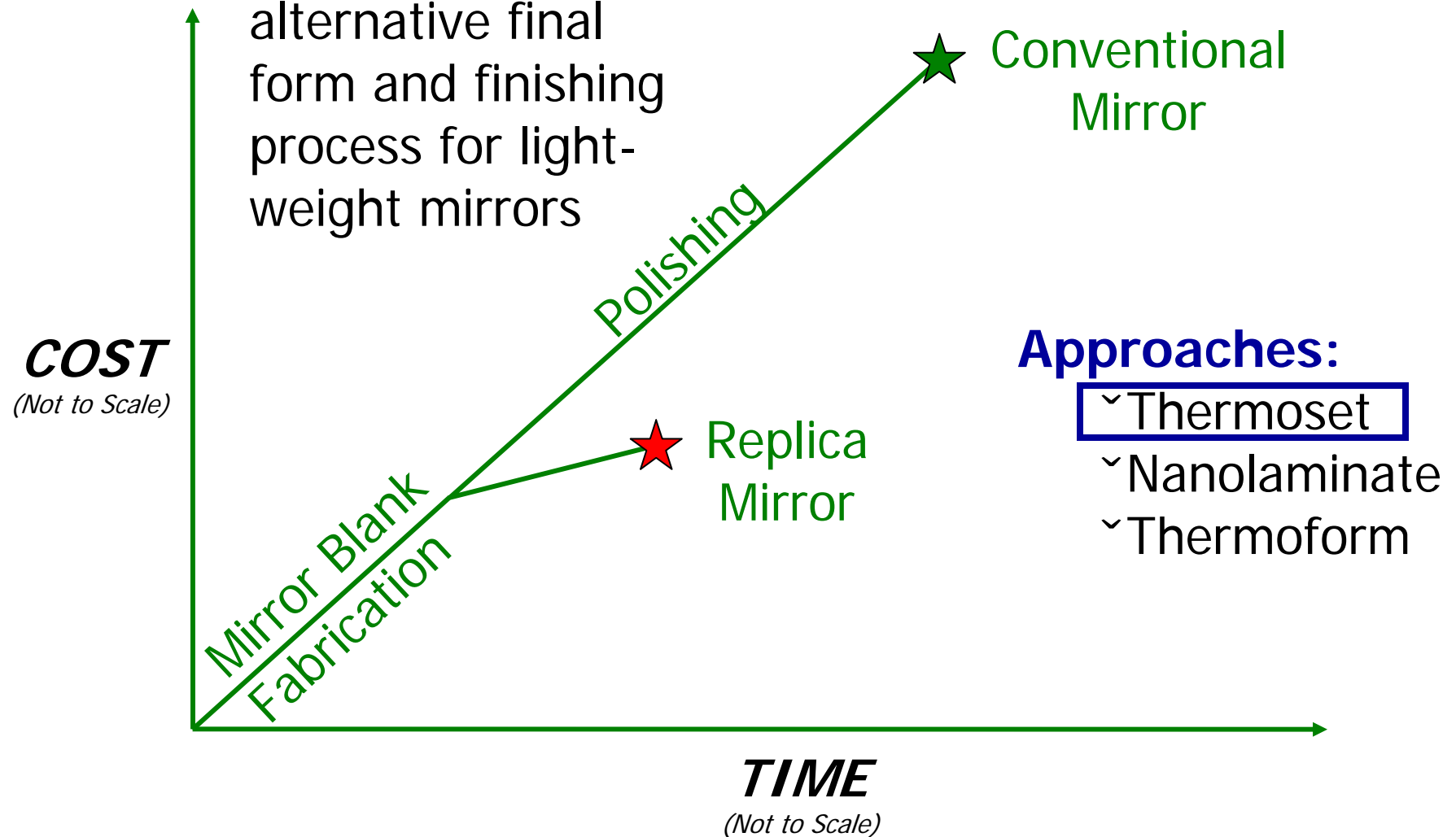


## Approaches:

- ~ Thermoset
- ~ Nanolaminate
- ~ Thermoform

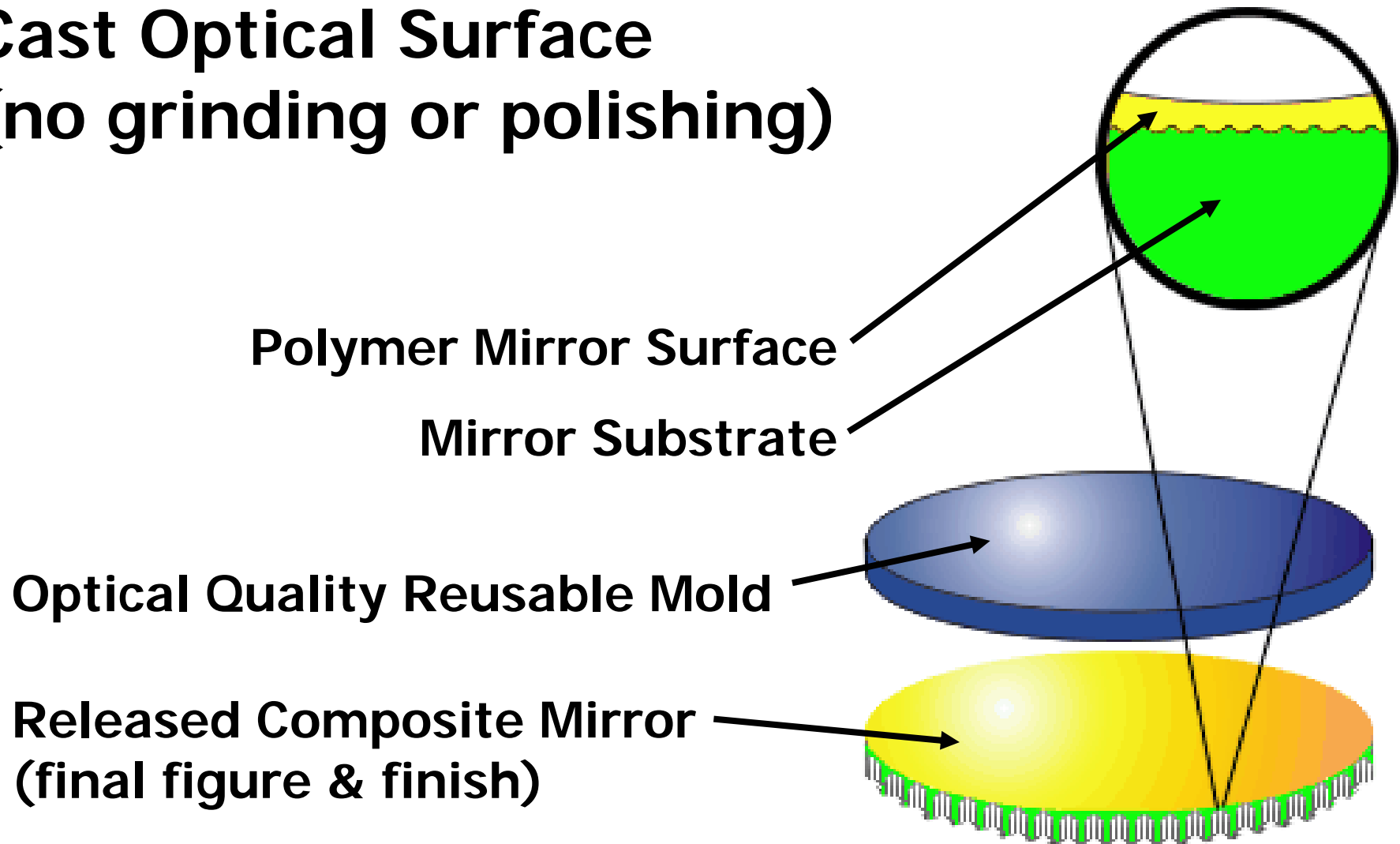
# PROGRAM INTRODUCTION: Replication Technology

**Goal:** Develop alternative final form and finishing process for light-weight mirrors



# PROGRAM INTRODUCTION: Thermoset Replica Concept

## Cast Optical Surface (no grinding or polishing)



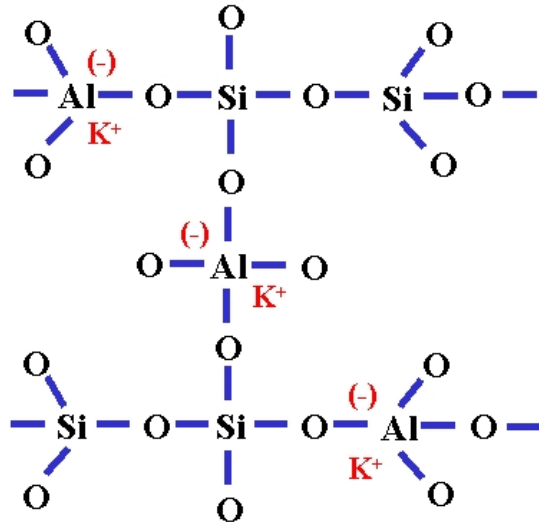
- Program Introduction
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## **PHASE I OBJECTIVES**

- 1. Formulate multi-component composites tailored for space-based mirrors**
- 2. Develop fabrication process**
- 3. Characterize candidate materials**
- 4. Assess candidates' feasibility for space-based mirrors**
- 5. Assess candidates' potential for mirror producibility**

- Program Introduction
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## PHASE I RESULTS: Sialyte™ Inorganic “Resin”



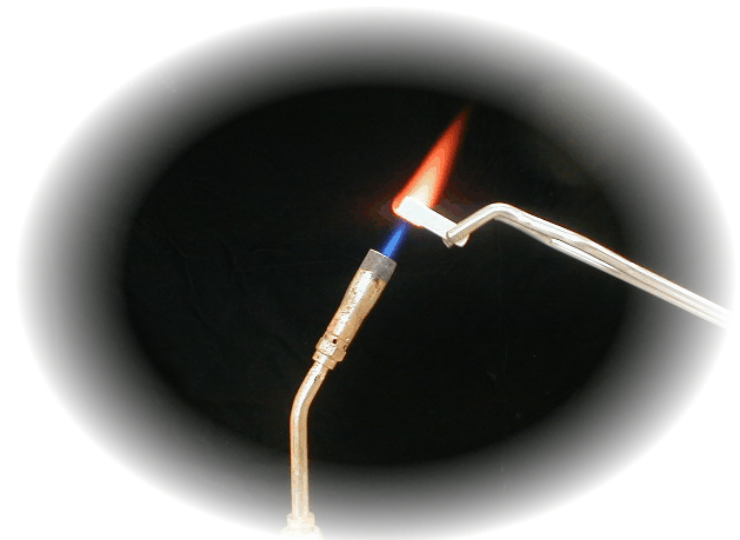
**poly(sialate-siloxo)**

### Attributes

- Inherently space compatible
- Lattice structure: high stiffness
- Operating temp: to ~900 °C  
bridges gap between organic resin  
and ceramics
- Low-temp process: fabrication savings

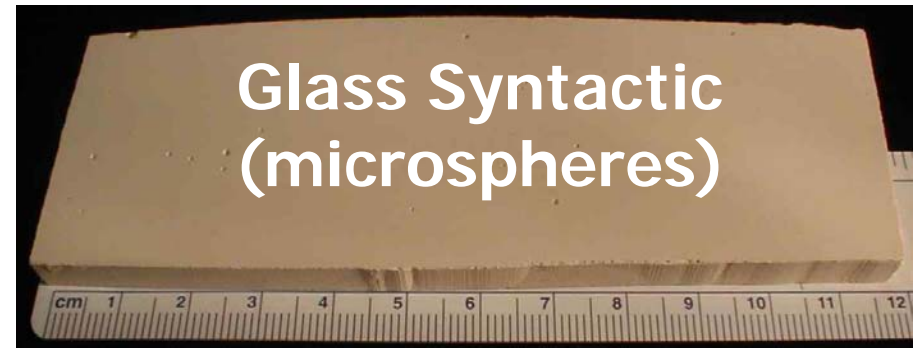
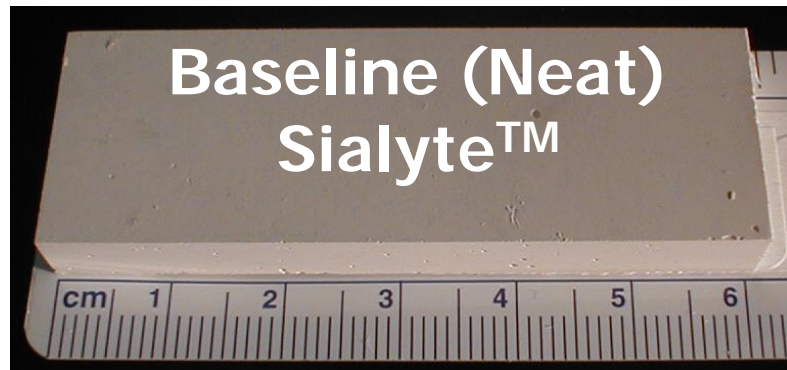
### Applications

- Space-based structures
- Propulsion components

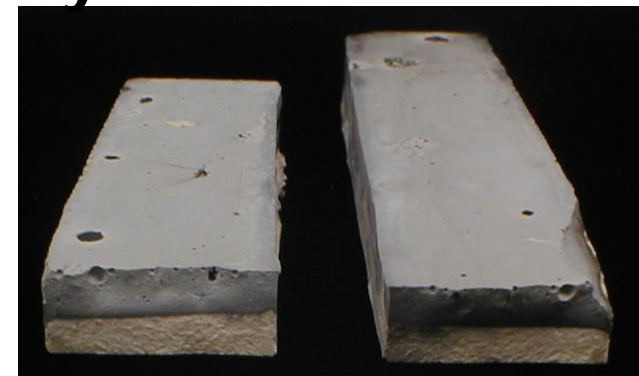




# PHASE I RESULTS: Representative Inorganic Candidates



## Carbon Nanofiber-Glass Syntactic Laminate



# PHASE I RESULTS: Sialyte™ Replica Mirror Coupon



## Fabrication

- Sialyte™ cast on optical flat
- Gold coating

## Finish

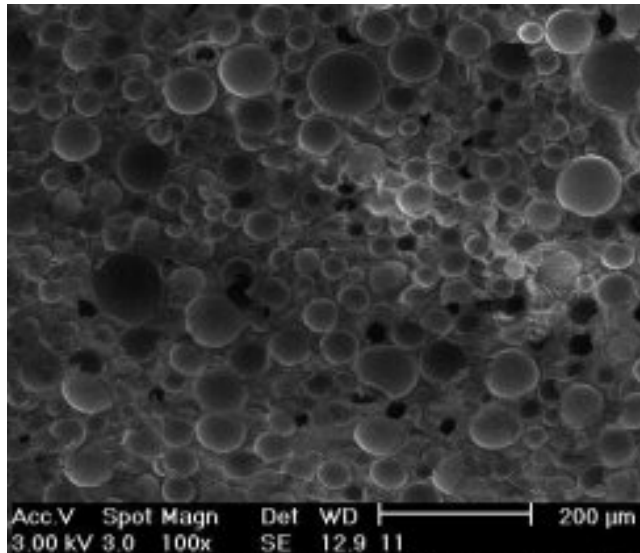
- Porous surface
- Roughness:
  - Best local: ~5 nm RMS (neat)
  - Best overall: ~8 nm RMS (ZrO<sub>2</sub> composite)

## **PHASE I RESULTS: Organic Materials**

- **Cyanate ester resin**
  - Demonstrated space compatible chemistry
  - Compatible with mature processes demonstrated with epoxy-based materials
    - Streamlines composite design
    - Streamlines process development
  - Formulation experience:  
Confidence in near term transition

# PHASE I RESULTS: High-Performance Syntactic Composite

**“Syntactic” = resin matrix + hollow microspheres**



## Attributes

- Low mass density: 0.55 g/cc
- High specific strength:  
126 MPa in compression
- Simple fabrication processes

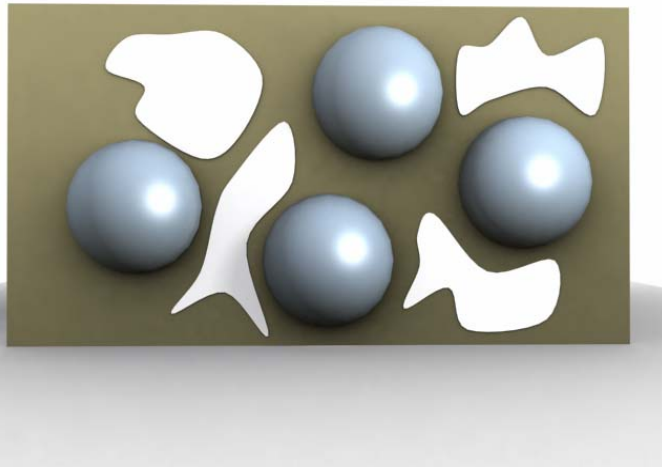
## Applications

- Lightweight structures
- Low dielectric structures
- High strength insulation

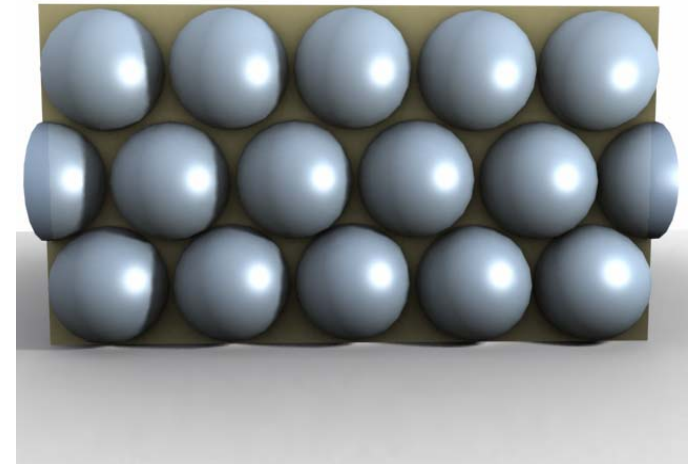


# PHASE I RESULTS: High-Performance Syntactic Composite

- **New fabrication technique**
  - Eliminates voids & increases microsphere loading
  - Improved material properties
    - Stronger
    - More uniform & more consistent



**Conventional  
Process**



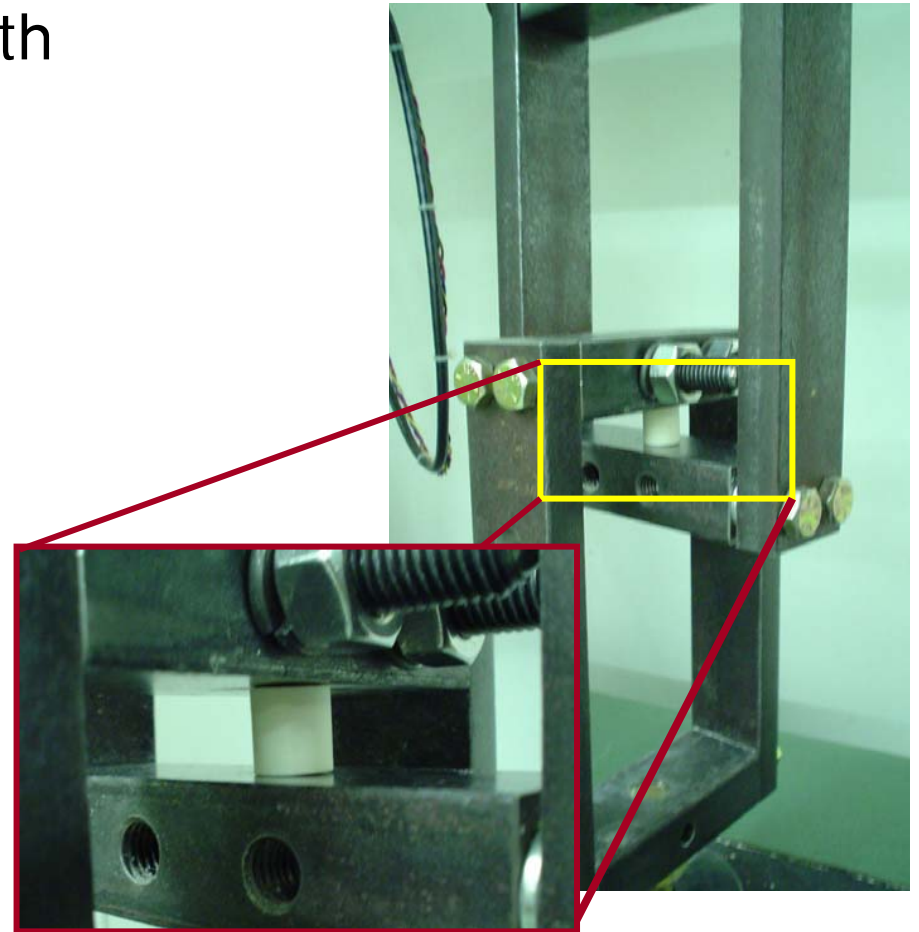
**CRG  
Process**



# PHASE I RESULTS: High-Performance Syntactic Composite

## Advantages for space mirrors

- Very high specific strength
- Low density



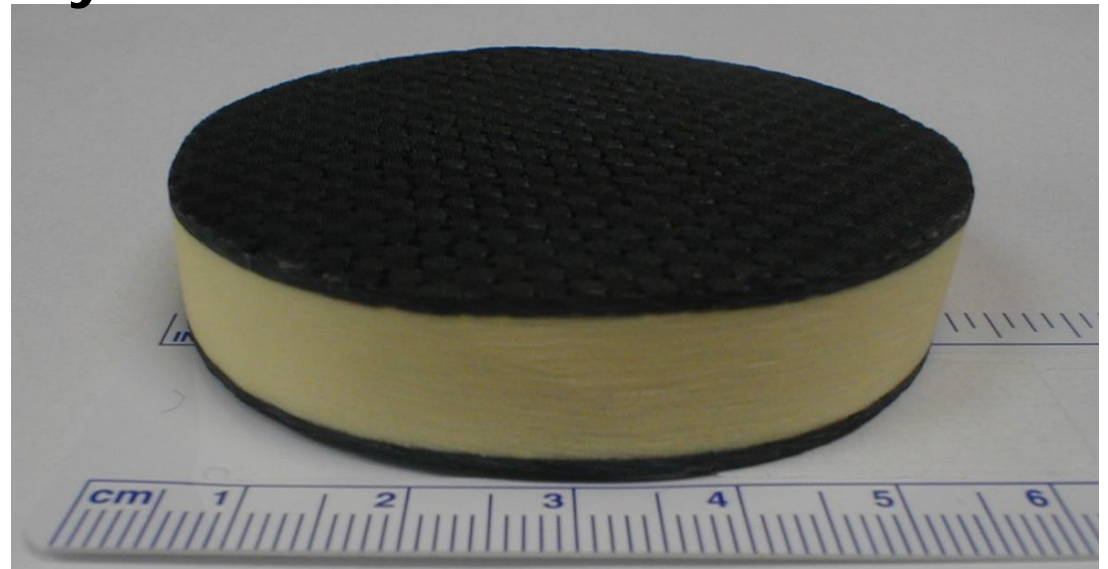
# PHASE I RESULTS: Representative Organic Candidates



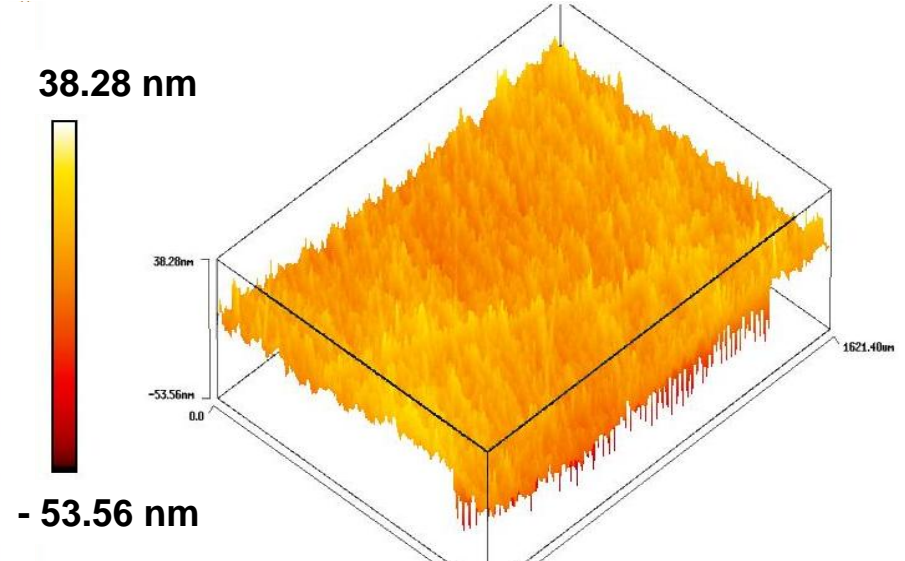
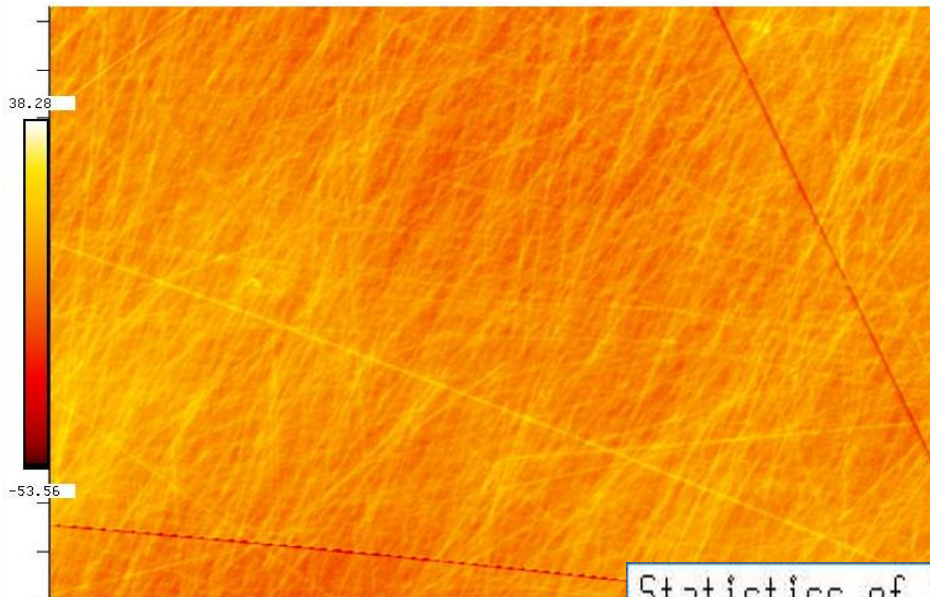
Carbon  
Nanofiber  
Reinforcement



Syntactic-Carbon Fiber Laminate



# PHASE I RESULTS: Cyanate Ester Surface Finish



Statistics of Surface: GK40-1A

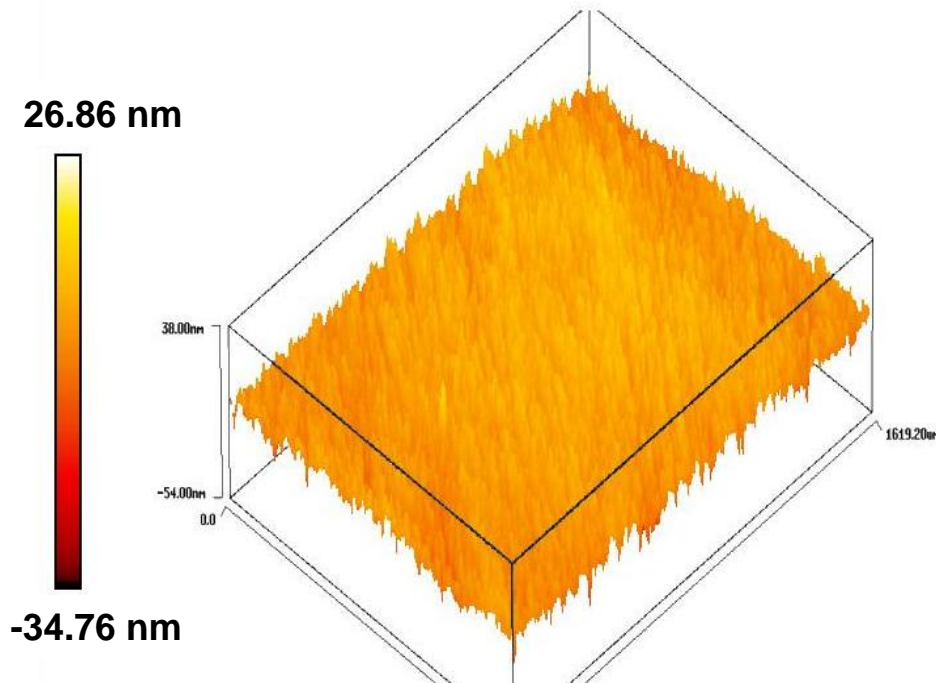
Rp: 38.28nm	Rq: 6.09nm	Area: 1621.40x1235.00um
Rv: -53.56nm	Ra: 4.74nm	Mag : 5.0
PV: 91.84nm	Rsk: 0.05	DATE: 06-14-2002
PT: 350071	Rku: 4.45	TIME: 10:03:38
Terms Subtracted: Tilt		



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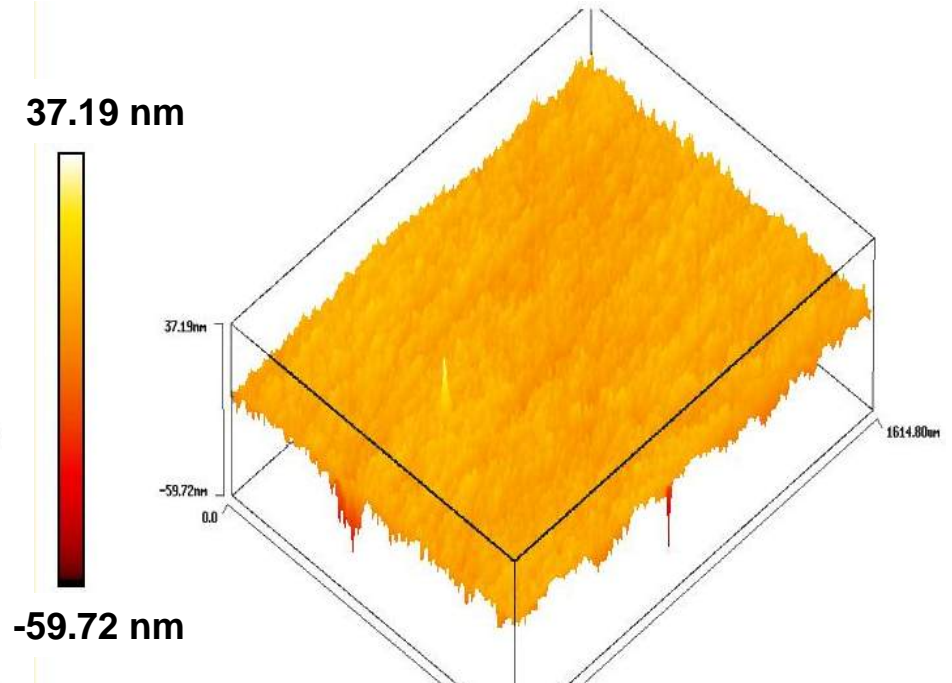


# PHASE I RESULTS: Mold Finish



Statistics of Surface: PUCK1-A			
Rp: 26.86nm	Rq: 5.41nm	Area: 1619.20x1229.80um	
Rv: -34.76nm	Ra: 4.30nm	Mag : 5.0	
PV: 61.62nm	Rsk: -0.19	DATE: 06-13-2002	
PT: 348127	Rku: 3.16	TIME: 09:29:50	
Terms Subtracted: Tilt			

Urxjkqhw#Ehiruh#F dwlqj ##  
8174#pp #JP V



Statistics of Surface: PUCK1-B			
Rp: 37.19nm	Rq: 4.61nm	Area: 1614.80x1229.80um	
Rv: -59.72nm	Ra: 3.59nm	Mag : 5.0	
PV: 96.91nm	Rsk: -0.65	DATE: 06-14-2002	
PT: 347182	Rku: 5.65	TIME: 11:56:13	
Terms Subtracted: Tilt			

Urxjkqhw#D ihu#F dwlqj ##  
7194#pp #JP V

## PHASE I RESULTS: Cyanate Ester - Syntactic Mirror

### OBJECTIVE

Demonstrate feasibility  
of replication approach

### RESULTS

#### Form

Slight curvature  
(due to cure shrinkage)

#### Finish

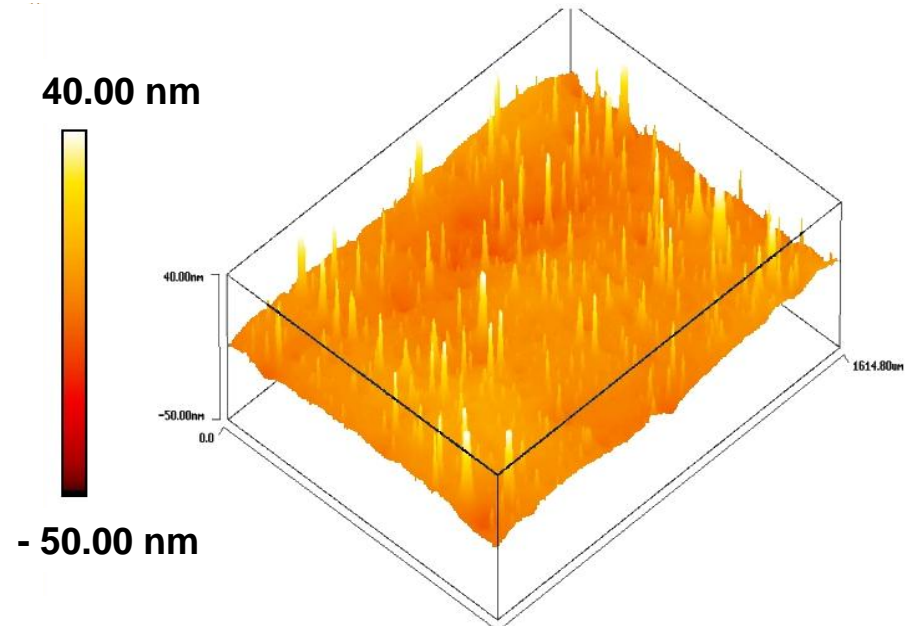
- Good mold replication
- Good reflective coating



#### Fabrication

- Good mold release
- Process development needed to improve figure replication
- Initial feasibility established

# PHASE I RESULTS: Cyanate Ester - Syntactic Mirror



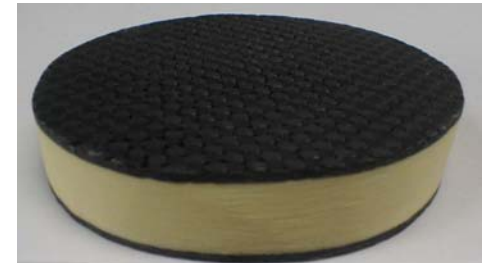
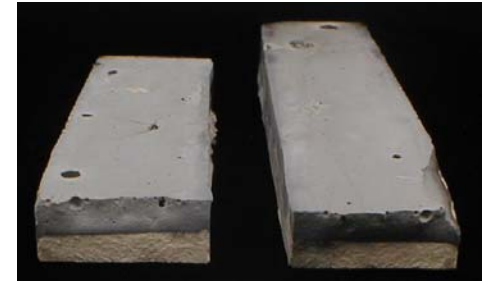
Statistics of Surface: GK84-B			
Rp: 166.37nm	Rq: 5.15nm	Area: 1614.80x1229.80um	
Rv: -32.51nm	Ra: 3.29nm	Mag : 5.0	
PV: 198.88nm	Rsk: 5.13	DATE: 09-05-2002	
PT: 347168	Rku: 91.18	TIME: 15:18:39	
Terms Subtracted: Tilt			

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8148#pp #JP V



## PHASE I RESULTS: Summary

- **Sialyte™ Inorganic Composites**
  - Multi-component composites feasible
  - Attributes promising for space mirrors
  - Need further development to reach transition
- **Cyanate Ester Organic Composites**
  - Multi-component composites feasible
  - Attributes demonstrated for replica mirrors for space-based optics
  - Ready for transition demonstration



- Program Introduction
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## **PHASE II PLAN: Technical Objectives**

- 1. Develop mature composite systems for replica mirrors (RM)**
  - a. Organic materials**
  - b. Inorganic materials**
- 2. Develop mature fabrication processes for RM**
- 3. Develop practical design methodology for RM**
- 4. Demonstrate a prototype composite RM**

## **PHASE II PLAN: Tasks in Progress**

- Optimize organic composites for replica mirrors
  - Materials
  - Processing
- Optimize inorganic composites for replica mirrors
  - Materials
  - Processing
- Mirror development
  - Structural design
  - Composite cutting
  - Fabrication (assembly and bonding)
  - Reflective surface
- Characterize materials and mirrors

- Program Introduction
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## PHASE II EARLY RESULTS: Inorganic Materials

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**Sialyte™ - Glass Fiber Composite**

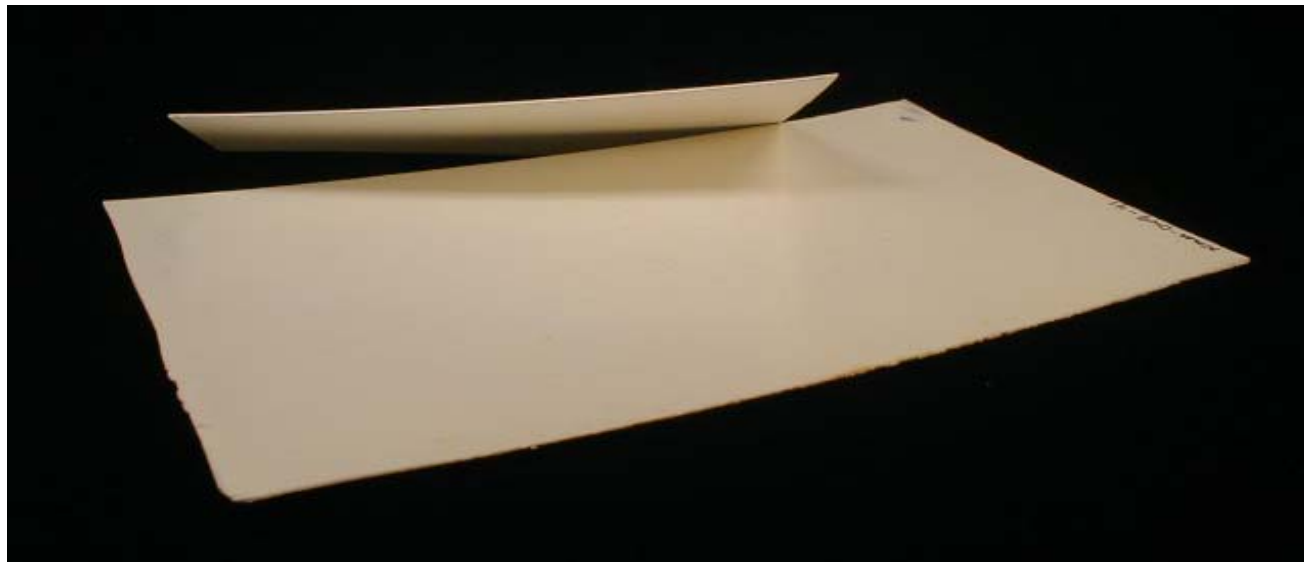
## PHASE II EARLY RESULTS: Inorganic Materials

- Processing improvements
  - Better control and repeatability over Phase I manual techniques
  - Developing techniques for complex shapes



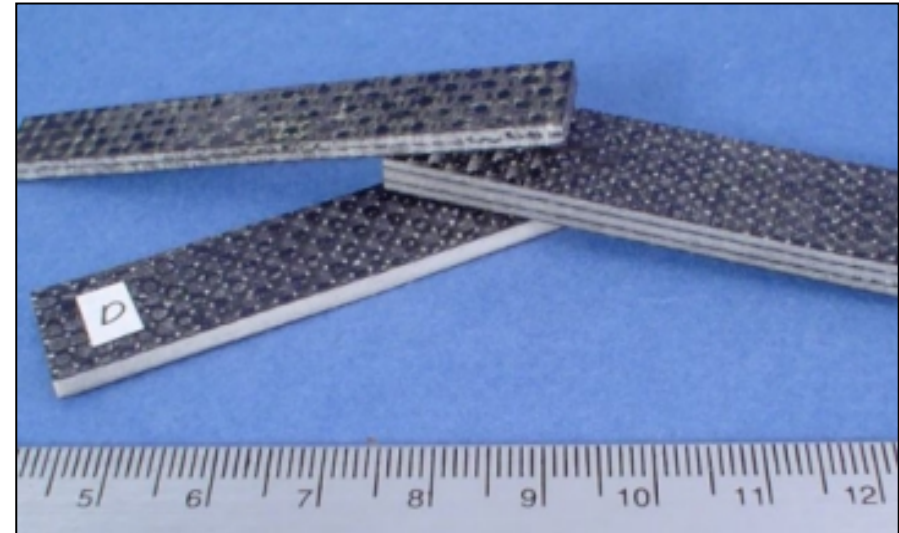
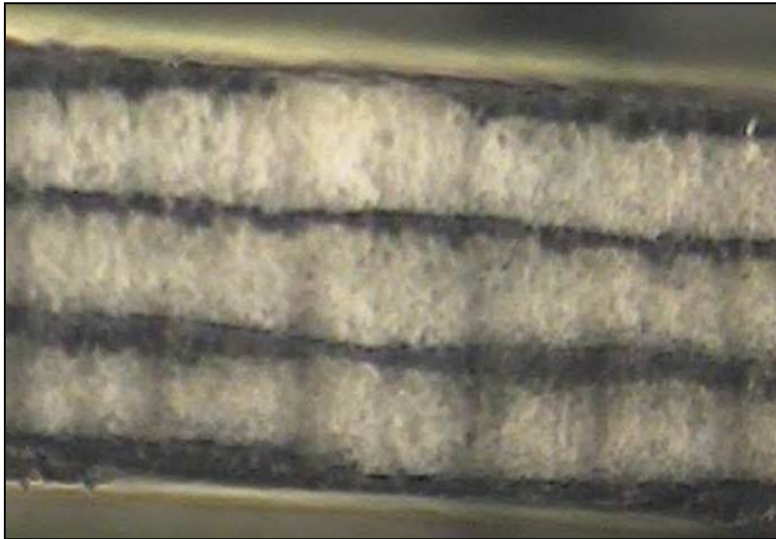
## PHASE II EARLY RESULTS: Organic Materials

- Syntactic materials
  - Process improvement & scale-up on track
    - Thinner & more uniform sheets
    - Larger sheets
  - Enabling technology for new **SynLam™** material



## PHASE II EARLY RESULTS: Organic Materials

- **SynLam™** developed for mirror structure:
  - Syn**tactic **Lam**inate composite
    - Syntactic sandwich cores
    - Fiber-reinforced face sheets





## PHASE II EARLY RESULTS: Organic Materials

- **SynLam™** Advantages
  - Lightweight
  - Maximum operating temperature 200 °C
  - 50% greater specific stiffness than conventional carbon fiber composite



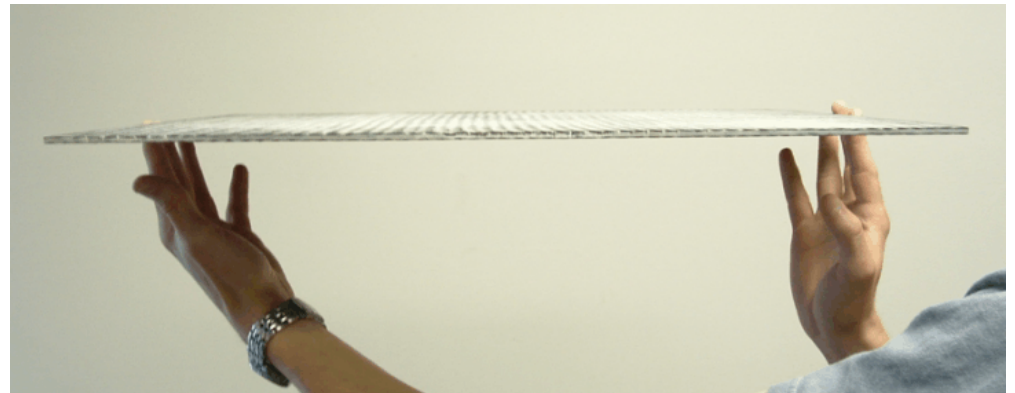
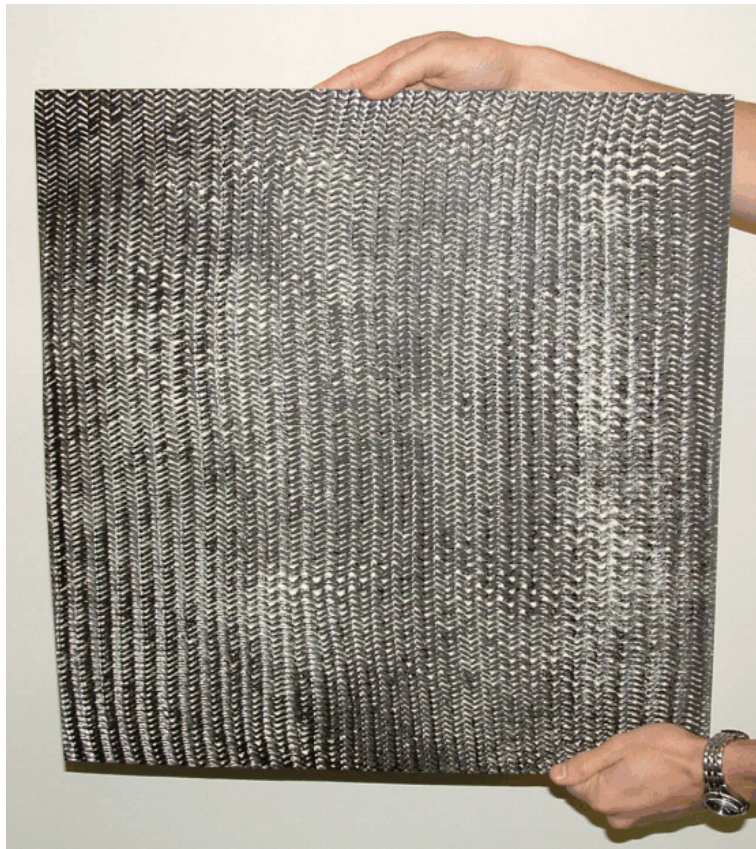
	Specific Flexural Strength	Specific Stiffness
<b>SynLam™</b>	371 MPa (54 kpsi)	33 GPa (4786 kpsi)
<b>Carbon Composite</b>	362 MPa (52 kpsi)	22 GPa (3191 kpsi)

## PHASE II EARLY RESULTS: Organic Materials

- SynLam™ Scaled Up:

From hand lay-up to automation

- Cyanate ester resin: 0.5 m x 0.5 m x 0.5 cm
- Epoxy resin: 0.9 m x 0.9 m x 0.5 cm



**Cyanate Ester SynLam™**

## PHASE II EARLY RESULTS: Organic Materials

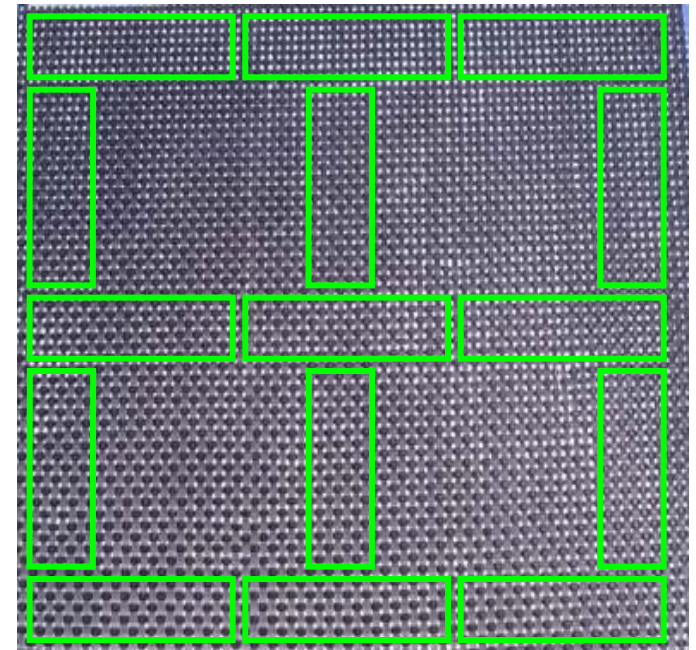
- Ongoing process development:  
Complex SynLam™ Structures





## PHASE II EARLY RESULTS: Organic Materials

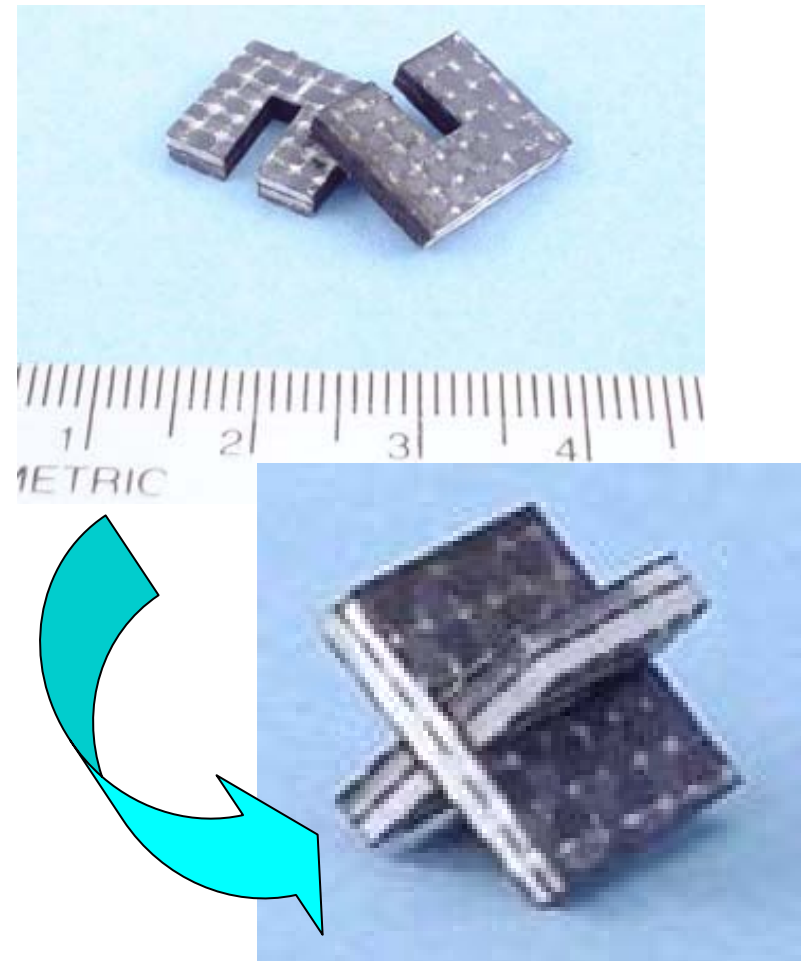
- Ongoing process development:  
    Quality Control
  - Current SynLam<sup>TM</sup> Uniformity  
(1/3 m scale sheets):
    - Density  $\pm 1\%$
    - Thickness  $\pm 2\%$
    - Specific modulus  $\pm 2\%$



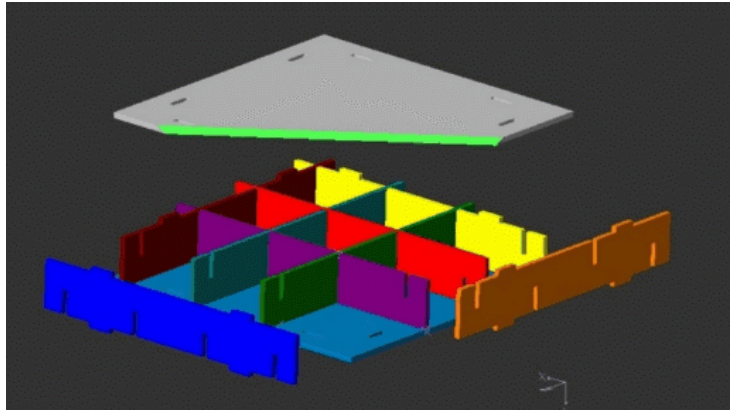


## PHASE II EARLY RESULTS: Organic Materials

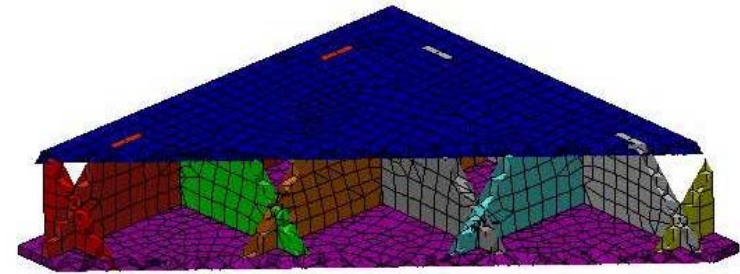
- Ongoing process development:  
Laser machining to cut SynLam™



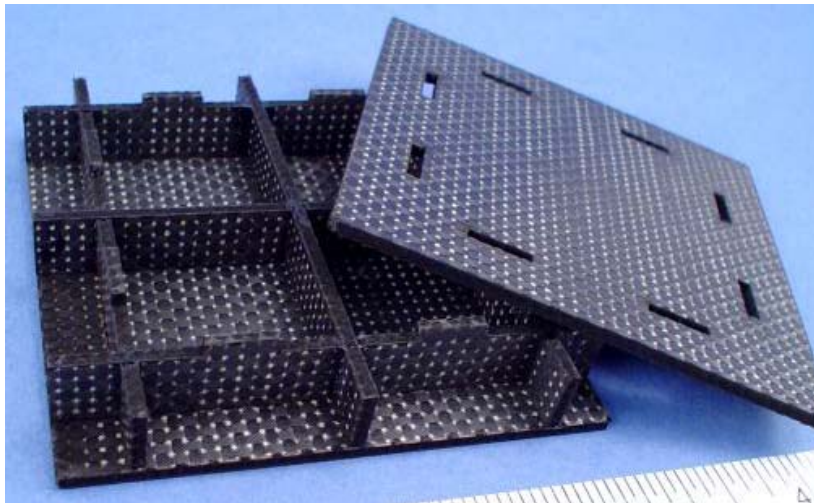
## PHASE II EARLY RESULTS: Mirror Development



Design



FE Analysis



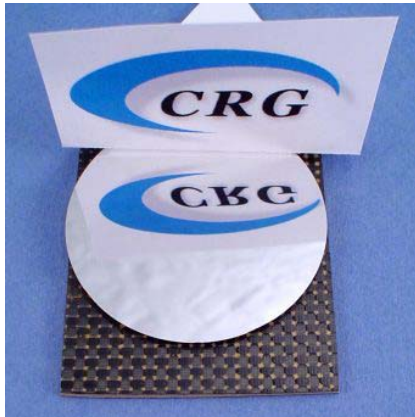
Fabrication

Coupon-Scale (8 cm across)

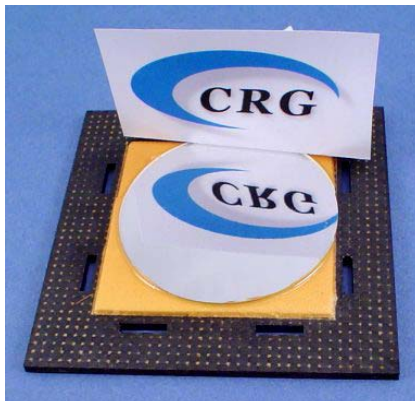
SynLam<sup>TM</sup> Structure

*Areal Density = 3.2 kg/m<sup>2</sup>*

## PHASE II EARLY RESULTS: Mirror Development

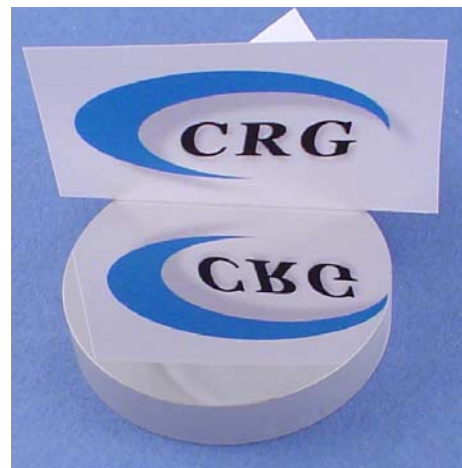


**Early Mirror  
(Print Through)**

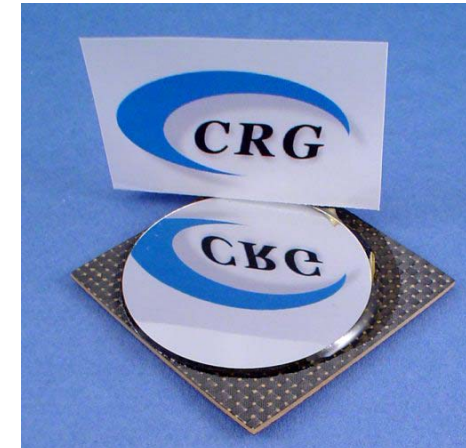


**Mirror w/Syntactic  
Buffer Layer**

**Cast Thermoset  
Replication**



**Mold with  
Release Coating**



**Mirror w/CE Resin  
Buffer Layer**

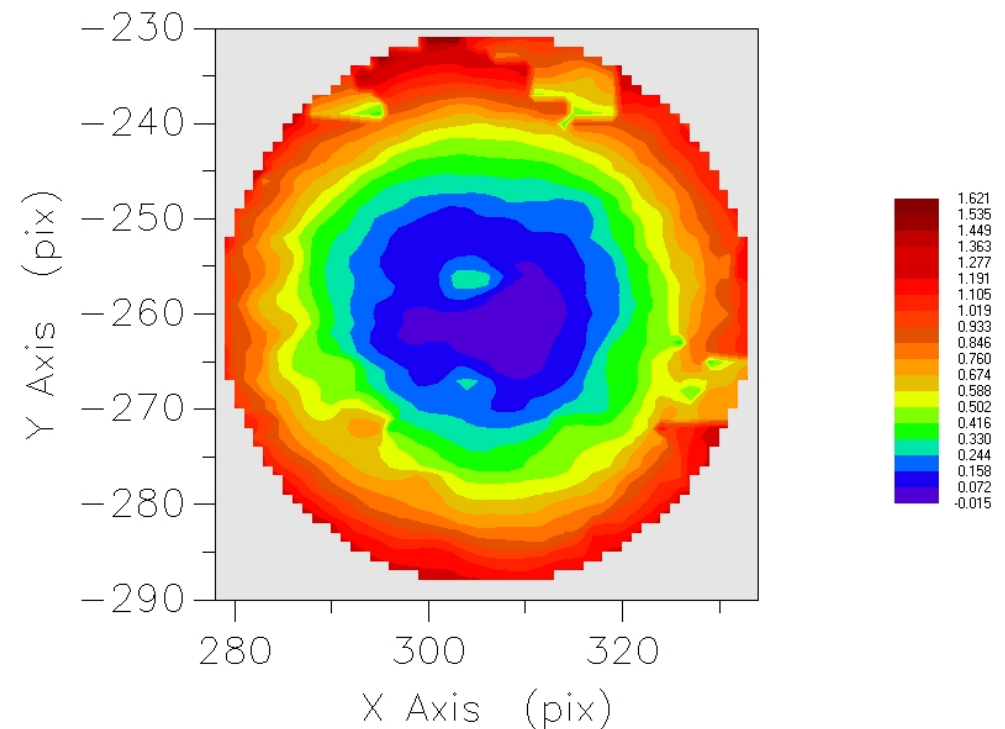
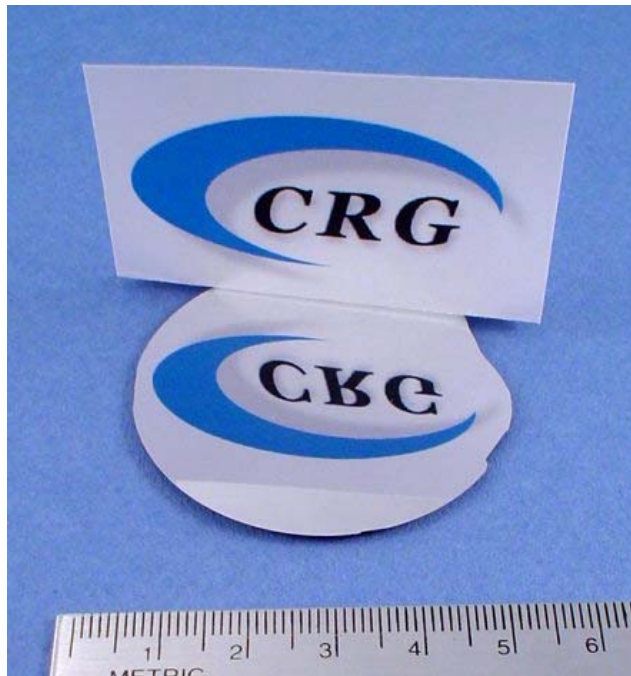


**Mirror on CE  
MWNT Composite**



# PHASE II EARLY RESULTS: Mirror Development

## Symmetric Figure



Range (PV) = 1.6358 waves, RMS = 0.3501 waves, Strehl = 0.0079  
Analysis Aper: Pos[ 306, 260] Size[ 57, 60]

## Development Progressing Rapidly

*Applying Tomorrow's Materials Today*

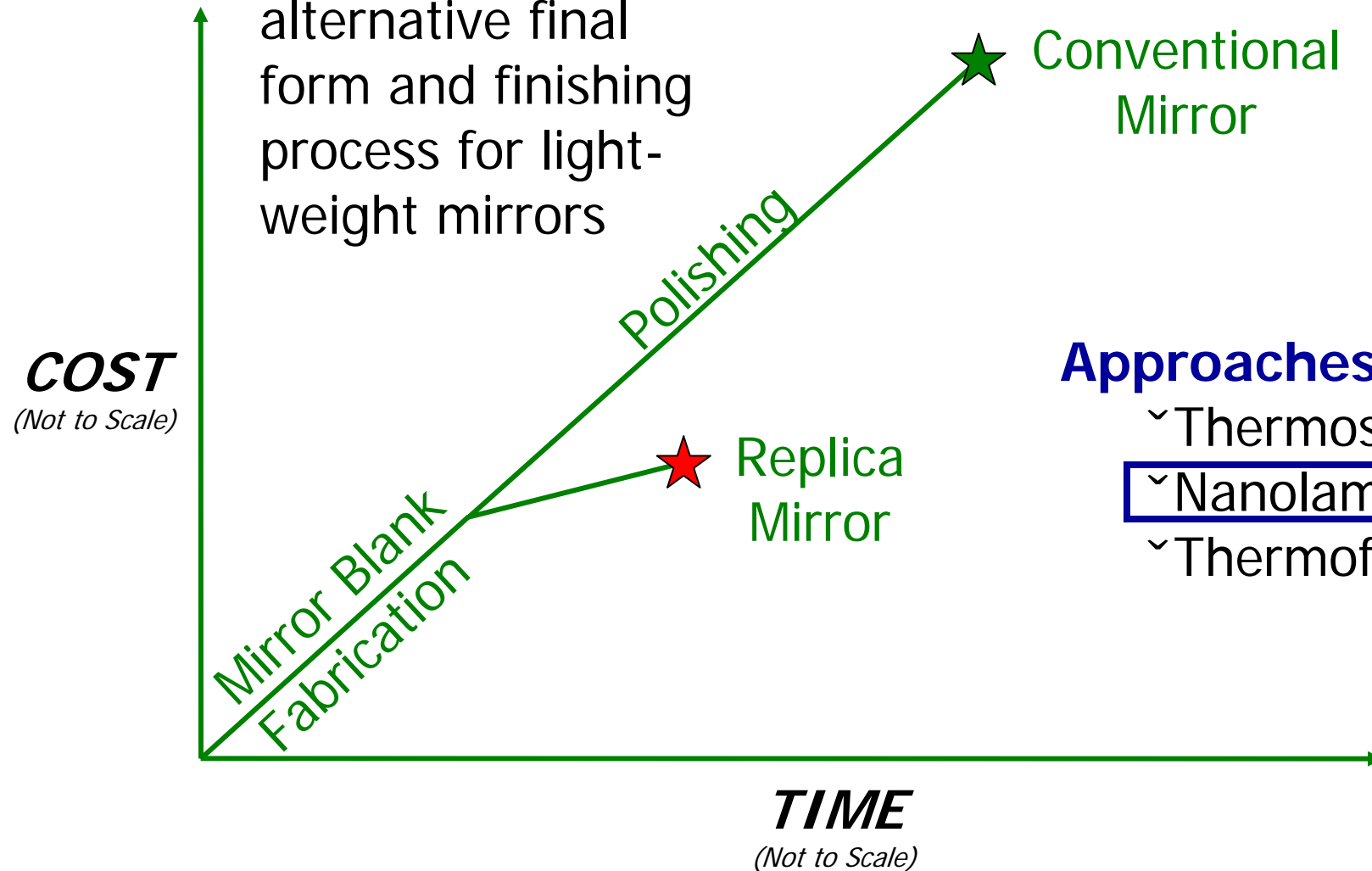
## **PHASE II EARLY RESULTS: Mirror Development**

- Current challenges
  - Improving optical surface
  - Interface between substrate and optical surface
  - Internal bonding of mirror structure  
(print-through issue)
  - Scale-up to prototype size

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# PROGRAM INTRODUCTION: Replication Technology

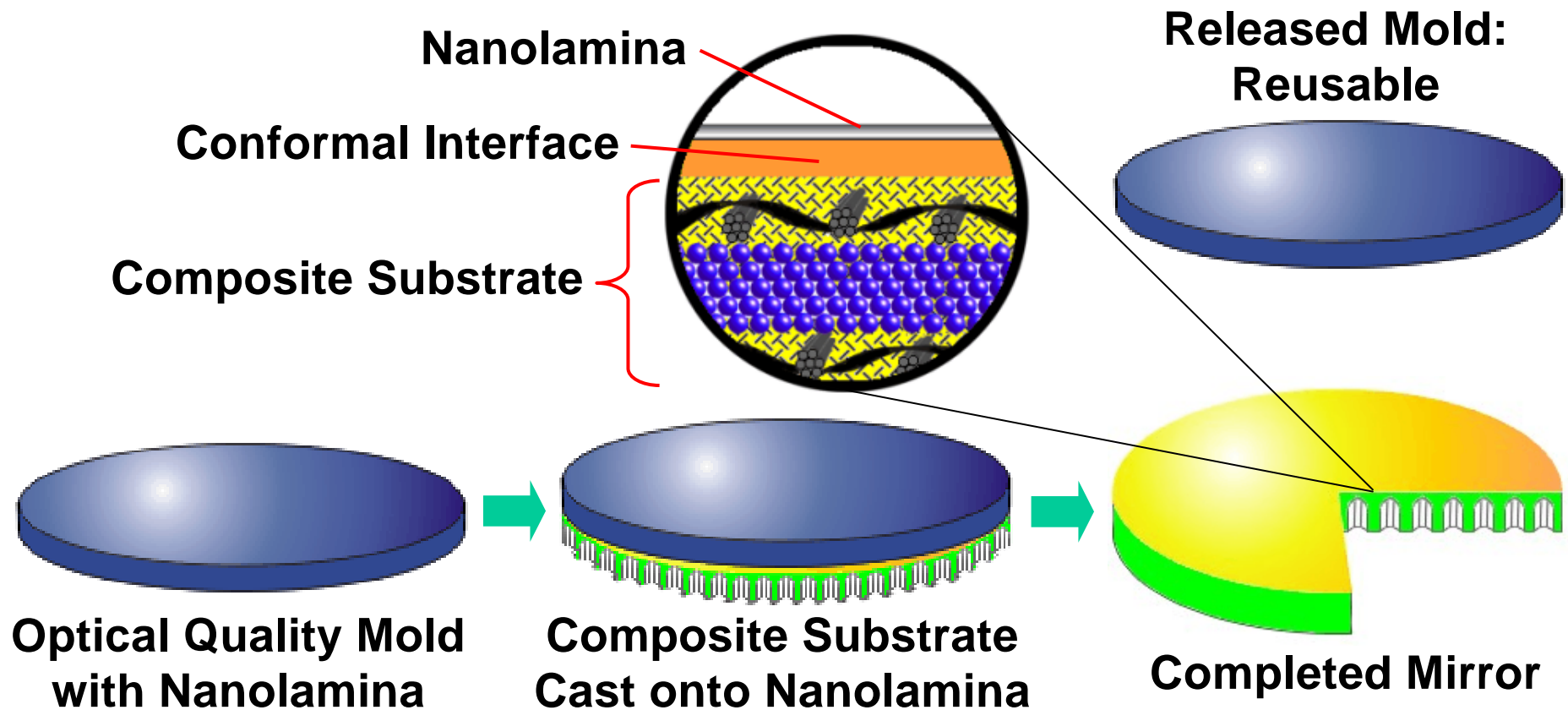
**Goal:** Develop alternative final form and finishing process for light-weight mirrors



## Approaches:

- ~ Thermoset
- ~ **Nanolaminate**
- ~ Thermoform

## PHASE II ENHANCEMENT: Nanolaminate Replica Concept





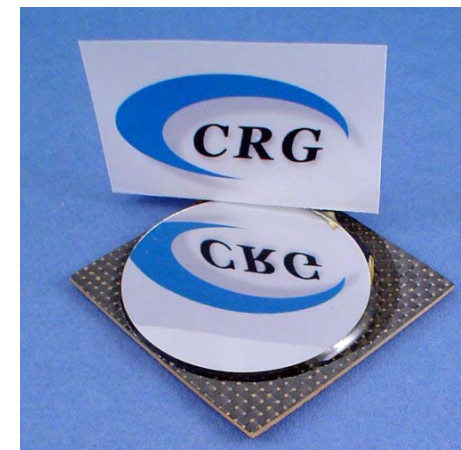
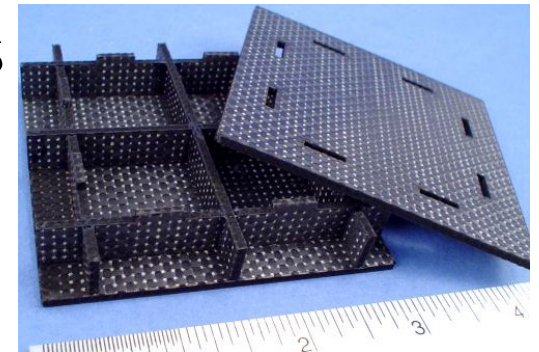
## **PHASE II ENHANCEMENT: Approach**

- Demonstrate Feasibility
  - Tailor materials for compatibility
  - Develop process for integrating lamina with structure
  - Evaluate thermal & optical performance
    - Quality of figure & finish at steady state
    - Stability during thermal cycling
- Demonstrate prototype replica mirror
  - 15 cm flat
  - Specular reflection
  - Cost-effective reusable tooling
- Effort commenced 15 June 2004

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## PHASE II SUMMARY: Early Results

- **Sialyte™ Inorganic Composites**
  - Appear promising for space mirrors
  - Development progressing
- **Cyanate Ester Organic Composites**
  - SynLam™ selected for structure
  - Optimizing & scaling up
- **Replication Processes**
  - Thermoset approach progressing
  - Nanolaminate approach pending





*Cornerstone Research Group, Inc.*

## PHASE II SUMMARY

### *Composite Replica Mirrors for Lightweight Space Optics*

- Operational Benefits
  - Reduced mirror areal density
  - Tougher & stronger mirrors
  - Reduced fabrication time & cost
- Potential Air Force Applications
  - Space-based imaging systems
  - Space-based directed energy systems
- Potential Commercial Applications
  - Commercial imaging systems (e.g., LANDSAT)
  - Consumer telescopes

